

AERODYNAMIC DRAG ON VARIOUS OBJECTS

In this lesson, we will:

- Look at tables of drag coefficient for various objects (geometric shapes, parachutes, trees, people, buildings, etc.)
- Do a “student-friendly” example problem – drag on a bicycle

Drag Coefficients on Objects of Various Shapes

See tables in the textbook, copied here. **Table 11-1: 2-D bodies**; **Table 11-2: 3-D bodies**.

TABLE 11-1 <i>2-D BODIES</i>																										
Drag coefficients C_D of various two-dimensional bodies for $Re > 10^4$ based on the frontal area $A = bD$, where b is the length in direction normal to the page (for use in the drag force relation $F_D = C_D A \rho V^2 / 2$ where V is the upstream velocity)																										
<p>Square rod</p> <p>Sharp corners: $C_D = 2.2$</p> <p>Round corners ($r/D = 0.2$): $C_D = 1.2$</p> <p><i>b >> D for 2-D body</i></p>	<p>Rectangular rod</p> <p>Sharp corners:</p> <table border="1"> <thead> <tr> <th>L/D</th> <th>C_D</th> </tr> </thead> <tbody> <tr><td>0.0*</td><td>1.9</td></tr> <tr><td>0.1</td><td>1.9</td></tr> <tr><td>0.5</td><td>2.5</td></tr> <tr><td>1.0</td><td>2.2</td></tr> <tr><td>2.0</td><td>1.7</td></tr> <tr><td>3.0</td><td>1.3</td></tr> </tbody> </table> <p>* Corresponds to thin plate</p> <p>Round front edge:</p> <table border="1"> <thead> <tr> <th>L/D</th> <th>C_D</th> </tr> </thead> <tbody> <tr><td>0.5</td><td>1.2</td></tr> <tr><td>1.0</td><td>0.9</td></tr> <tr><td>2.0</td><td>0.7</td></tr> <tr><td>4.0</td><td>0.7</td></tr> </tbody> </table>	L/D	C_D	0.0*	1.9	0.1	1.9	0.5	2.5	1.0	2.2	2.0	1.7	3.0	1.3	L/D	C_D	0.5	1.2	1.0	0.9	2.0	0.7	4.0	0.7	
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<p>Circular rod (cylinder)</p> <p>Laminar: $C_D = 1.2$</p> <p>Turbulent: $C_D = 0.3$</p>	<p>Elliptical rod</p> <table border="1"> <thead> <tr> <th rowspan="2">L/D</th> <th colspan="2">C_D</th> </tr> <tr> <th>Laminar</th> <th>Turbulent</th> </tr> </thead> <tbody> <tr><td>2</td><td>0.60</td><td>0.20</td></tr> <tr><td>4</td><td>0.35</td><td>0.15</td></tr> <tr><td>8</td><td>0.25</td><td>0.10</td></tr> </tbody> </table>	L/D	C_D		Laminar	Turbulent	2	0.60	0.20	4	0.35	0.15	8	0.25	0.10											
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<p>Equilateral triangular rod</p> <p>$C_D = 1.5$</p> <p>$C_D = 2.0$</p>	<p>Semicircular shell</p> <p>$C_D = 2.3$</p> <p>$C_D = 1.2$</p>	<p>Semicircular rod</p> <p>$C_D = 1.2$</p> <p>$C_D = 1.7$</p>																								

Tables from Çengel and Cimbala, Ed. 4.

TABLE 11-2 3-D BODIES

Representative drag coefficients C_D for various three-dimensional bodies based on the frontal area for $Re > 10^4$ unless stated otherwise (for use in the drag force relation $F_D = C_D A \rho V^2 / 2$ where V is the upstream velocity)

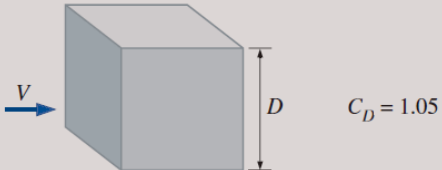
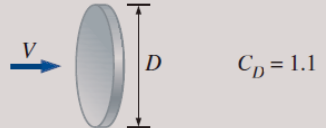
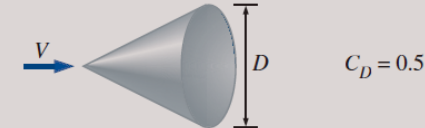
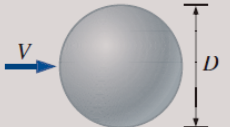
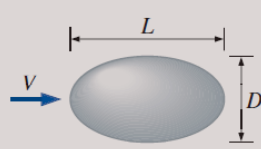
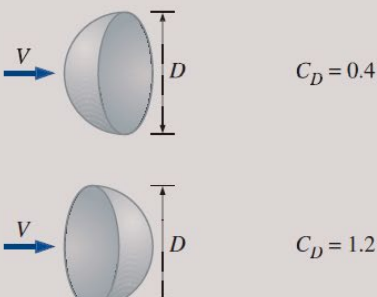
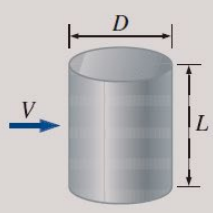
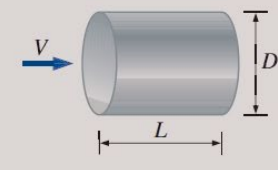

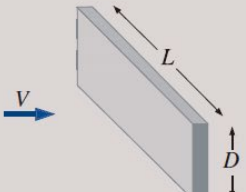

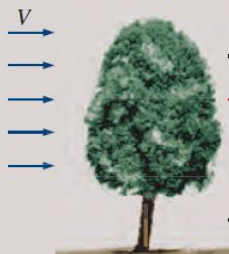





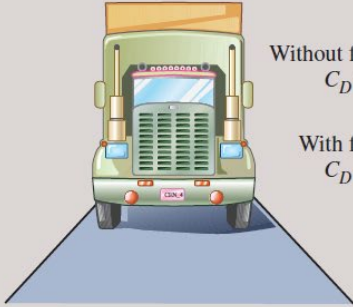



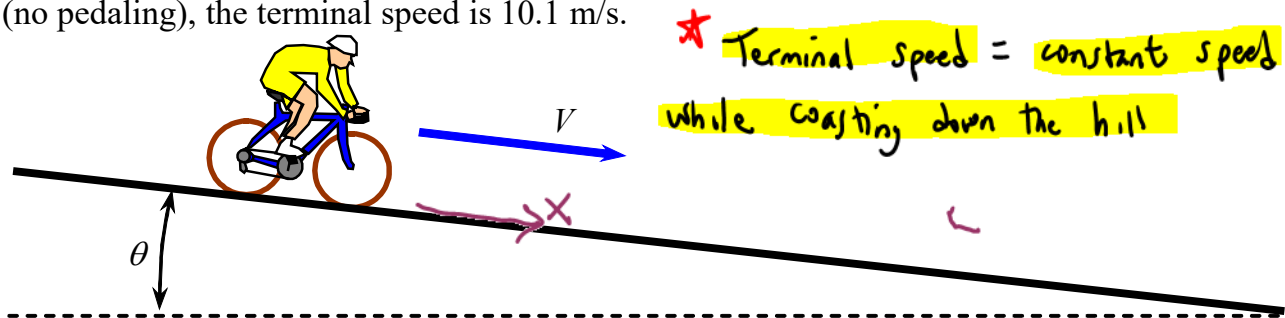
<p>Cube, $A = D^2$</p>  <p>$C_D = 1.05$</p>	<p>Thin circular disk, $A = \pi D^2/4$</p>  <p>$C_D = 1.1$</p>	<p>Cone (for $\theta = 30^\circ$), $A = \pi D^2/4$</p>  <p>$C_D = 0.5$</p>																										
<p>Sphere, $A = \pi D^2/4$</p>  <p>Laminar: $Re \leq 2 \times 10^5$ $C_D = 0.5$ Turbulent: $Re \geq 2 \times 10^6$ $C_D = 0.2$</p> <p>See Fig. 11-36 for C_D vs. Re for smooth and rough spheres.</p>	<p>Ellipsoid, $A = \pi D^2/4$</p> 	<table border="1"> <thead> <tr> <th rowspan="2">L/D</th> <th colspan="2">C_D</th> </tr> <tr> <th>Laminar $Re \leq 2 \times 10^5$</th> <th>Turbulent $Re \geq 2 \times 10^6$</th> </tr> </thead> <tbody> <tr> <td>0.75</td> <td>0.5</td> <td>0.2</td> </tr> <tr> <td>1</td> <td>0.5</td> <td>0.2</td> </tr> <tr> <td>2</td> <td>0.3</td> <td>0.1</td> </tr> <tr> <td>4</td> <td>0.3</td> <td>0.1</td> </tr> <tr> <td>8</td> <td>0.2</td> <td>0.1</td> </tr> </tbody> </table>	L/D	C_D		Laminar $Re \leq 2 \times 10^5$	Turbulent $Re \geq 2 \times 10^6$	0.75	0.5	0.2	1	0.5	0.2	2	0.3	0.1	4	0.3	0.1	8	0.2	0.1						
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<p>Hemisphere, $A = \pi D^2/4$</p>  <p>$C_D = 0.4$</p> <p>$C_D = 1.2$</p>	<p>Finite cylinder, vertical, $A = LD$</p>  <table border="1"> <thead> <tr> <th>L/D</th> <th>C_D</th> </tr> </thead> <tbody> <tr> <td>1</td> <td>0.6</td> </tr> <tr> <td>2</td> <td>0.7</td> </tr> <tr> <td>5</td> <td>0.8</td> </tr> <tr> <td>10</td> <td>0.9</td> </tr> <tr> <td>40</td> <td>1.0</td> </tr> <tr> <td>∞</td> <td>1.2</td> </tr> </tbody> </table> <p>Values are for laminar flow ($Re \leq 2 \times 10^5$)</p>	L/D	C_D	1	0.6	2	0.7	5	0.8	10	0.9	40	1.0	∞	1.2	<p>Finite cylinder, horizontal, $A = \pi D^2/4$</p>  <table border="1"> <thead> <tr> <th>L/D</th> <th>C_D</th> </tr> </thead> <tbody> <tr> <td>0.5</td> <td>1.1</td> </tr> <tr> <td>1</td> <td>0.9</td> </tr> <tr> <td>2</td> <td>0.9</td> </tr> <tr> <td>4</td> <td>0.9</td> </tr> <tr> <td>8</td> <td>1.0</td> </tr> </tbody> </table>	L/D	C_D	0.5	1.1	1	0.9	2	0.9	4	0.9	8	1.0
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<p>Streamlined body, $A = \pi D^2/4$</p>  <p>$C_D = 0.04$</p> <hr/> <p>Rectangular plate, $A = LD$</p>  <p>$C_D = 1.10 + 0.02 (L/D + D/L)$ for $1/30 < (L/D) < 30$</p>	<p>Parachute, $A = \pi D^2/4$</p>  <p>$C_D = 1.3$</p>	<p>Tree, $A = \text{frontal area}$</p> <p>$A = \text{frontal area}$</p>  <table border="1"> <thead> <tr> <th>$V, \text{ m/s}$</th> <th>C_D</th> </tr> </thead> <tbody> <tr> <td>10</td> <td>0.4–1.2</td> </tr> <tr> <td>20</td> <td>0.3–1.0</td> </tr> <tr> <td>30</td> <td>0.2–0.7</td> </tr> </tbody> </table>	$V, \text{ m/s}$	C_D	10	0.4–1.2	20	0.3–1.0	30	0.2–0.7																		
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TABLE 11-2 (Continued)

<p>Person (average)</p>  <p>Standing: $C_D A = 9 \text{ ft}^2 = 0.84 \text{ m}^2$ Sitting: $C_D A = 6 \text{ ft}^2 = 0.56 \text{ m}^2$</p>	<p>Bikes</p>  <p>Upright: $A = 5.5 \text{ ft}^2 = 0.51 \text{ m}^2$ $C_D = 1.1$</p>  <p>Racing: $A = 3.9 \text{ ft}^2 = 0.36 \text{ m}^2$ $C_D = 0.9$</p>	<p>$C_D = 0.9$ $C_D = 0.5$</p>  <p>Drafting: $A = 3.9 \text{ ft}^2 = 0.36 \text{ m}^2$ $C_D = 0.50$</p>  <p>With fairing: $A = 5.0 \text{ ft}^2 = 0.46 \text{ m}^2$ $C_D = 0.12$</p>
<p>Semitrailer, $A =$ frontal area</p>  <p>Without fairing: $C_D = 0.96$</p> <p>With fairing: $C_D = 0.76$</p>	<p>Automotive, $A =$ frontal area</p>  <p>Minivan: $C_D = 0.4$</p>  <p>Passenger car or sports car: $C_D = 0.3$</p>	<p>High-rise buildings, $A =$ frontal area</p> <p>$C_D \approx 1.0$ to 1.4</p> 

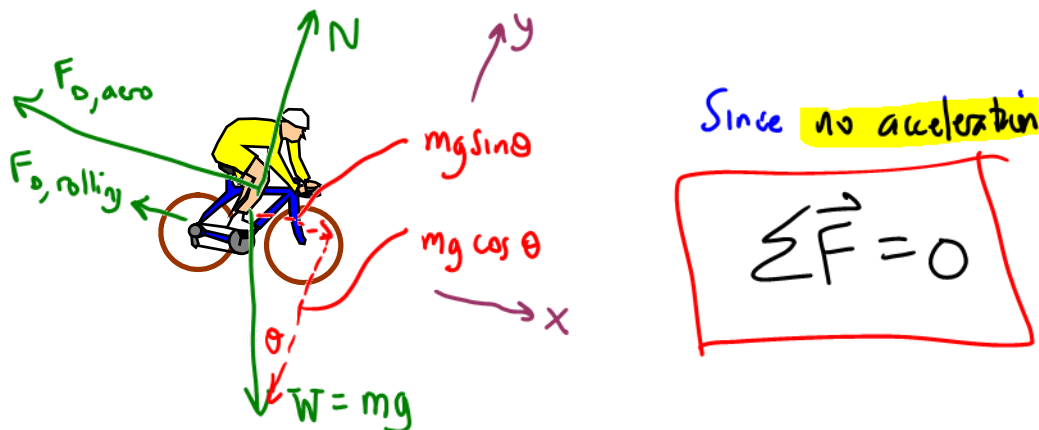
Example – Drag on a Bicycle Rolling Down a Hill

Given: A person coasts a bicycle down a long hill with a slope of 5° in order to measure the drag area of the bike and rider. The mass of the bike is 7.0 kg , the mass of the rider is 70.0 kg , and the rolling resistance of the bike is measured separately – it is 19.0 N . When the rider coasts down the hill (no pedaling), the terminal speed is 10.1 m/s .



(a) To do: Calculate the drag area $C_D A$ of the rider/bicycle combination in m^2 .

Solution: First draw a free-body diagram of the bicycle and rider, showing all forces acting.



$$\sum F_x = 0 \quad ; \quad \sum F_y = 0$$

$$mg \sin \theta - F_{D, \text{rolling}} - F_{D, \text{aero}} = 0$$

$$mg \sin \theta = F_{D, \text{rolling}} + \frac{1}{2} \rho V^2 C_D A$$

This is what we want to calculate

$$C_D A = \text{Drag area} = \frac{mg \sin \theta - F_{D, \text{rolling}}}{\frac{1}{2} \rho V^2}$$

answer in variable form

Plug in the values:

$$C_D A = \frac{(70 + 7) \text{ kg} (9.807 \frac{\text{m}}{\text{s}^2}) \left(\frac{\text{N} \cdot \text{s}^2}{\text{kg} \cdot \text{m}} \right) \sin 5^\circ - 19.0 \text{ N}}{\frac{1}{2} (1.204 \frac{\text{kg}}{\text{m}^3}) (10.1 \frac{\text{m}}{\text{s}})^2 \left(\frac{\text{N} \cdot \text{s}^2}{\text{kg} \cdot \text{m}} \right)} = 0.762 \text{ m}^2$$

$$C_D A = 0.762 \text{ m}^2$$

(b) **To do:** Calculate how much power in Watts (to the wheel) it would take for the person to ride this bike on a level road at the same speed (10.1 m/s).

Solution: Use the same equation we had in a previous lesson for automobiles:

$$\dot{W} = \underbrace{\mu_{\text{rolling}} W V}_{F_{D, \text{rolling}}} + \frac{1}{2} \rho V^3 \underbrace{C_D A}_{\text{Does not change with speed}}$$

numbers:

$$\dot{W} = \left[(19.0 \text{ N}) (10.1 \frac{\text{m}}{\text{s}}) + \frac{1}{2} (1.204 \frac{\text{kg}}{\text{m}^3}) (10.1 \frac{\text{m}}{\text{s}})^3 (0.76233 \text{ m}^2) \left(\frac{\text{N} \cdot \text{s}^2}{\text{kg} \cdot \text{m}} \right) \right] \left(\frac{\text{W} \cdot \text{s}}{\text{N} \cdot \text{m}} \right)$$

$$\dot{W} = 665. \text{ W} \quad \approx \underline{\underline{0.89 \text{ hp} !}}$$