

**Appendix A.10** Critical temperatures and critical pressures for common toxicants (abstracted from CRC Press, 1975 and ACGIH Ventilation Manual, 1988).

<b>name</b>	<b>formula</b>	<b>T<sub>c</sub> (°C)</b>	<b>P<sub>c</sub> (atm)</b>
acetaldehyde	C <sub>4</sub> H <sub>4</sub> O	187.8	54.7
acetic acid	C <sub>2</sub> H <sub>4</sub> O <sub>2</sub>	321.6	57.1
acetone	C <sub>3</sub> H <sub>6</sub> O	235.5	47
acetonitrile	C <sub>2</sub> H <sub>3</sub> N	274.7	47.7
aniline	C <sub>6</sub> H <sub>7</sub> N	425.6	52.3
benzene	C <sub>6</sub> H <sub>6</sub>	288.9	48.6
benzyl chloride	C <sub>6</sub> H <sub>5</sub> Cl	359.2	44.6
boron trifluoride	BF <sub>3</sub>	-12.3	49.2
carbon disulfide	CS <sub>2</sub>	279	78
carbon tetrachloride	CCl <sub>4</sub>	283.4	45.6
diethylamine	C <sub>4</sub> H <sub>11</sub> N	223.3	36.6
dimethylamine	C <sub>2</sub> H <sub>7</sub> N	164.6	52.4
ethylene oxide	C <sub>2</sub> H <sub>5</sub> O	195.8	71
hydrogen chloride	HCl	51.4	82.1
hydrogen cyanide	HCN	183.5	48.9
hydrogen sulfide	H <sub>2</sub> S	100.4	88.9
methyl alcohol	CH <sub>4</sub> O	240	78.5
methylamine	CH <sub>5</sub> N	156.9	40.2
methylene chloride	CH <sub>2</sub> Cl <sub>2</sub>	237	60
methyl mercaptan	CH <sub>4</sub> S	196.8	71.4
naphthalene	C <sub>10</sub> H <sub>8</sub>	474.8	40.6
nitric oxide	NO	-93	64
ozone	O <sub>3</sub>	-5.2	67
phenol	C <sub>6</sub> H <sub>6</sub> O	421.1	60.5
propylene oxide	C <sub>3</sub> H <sub>6</sub> O	209	48.6
styrene	C <sub>8</sub> H <sub>8</sub>	374.4	39.4
triethylamine	C <sub>6</sub> H <sub>15</sub> N	258.9	30
toluene	C <sub>7</sub> H <sub>8</sub>	320.8	41.6

**Appendix A.11** Thermophysical properties of air (abstracted from Schmidt et al., 1984).

T (K)	$\rho$ (kg/m <sup>3</sup> )	$C_p$ (kJ/kg K)	$\mu$ (N s/m <sup>2</sup> )	$\nu$ (m <sup>2</sup> /s)	$\kappa$ (W/m K)	$\alpha$ (m <sup>2</sup> /s)	Pr
200	1.7458	1.007	132.5 x 10 <sup>-7</sup>	7.590 x 10 <sup>-6</sup>	18.1 x 10 <sup>-3</sup>	10.3 x 10 <sup>-6</sup>	0.737
250	1.3947	1.006	159.6 x 10 <sup>-7</sup>	11.44 x 10 <sup>-6</sup>	22.3 x 10 <sup>-3</sup>	15.9 x 10 <sup>-6</sup>	0.720
300	1.1614	1.007	184.6 x 10 <sup>-7</sup>	15.89 x 10 <sup>-6</sup>	26.3 x 10 <sup>-3</sup>	22.5 x 10 <sup>-6</sup>	0.707
350	0.9950	1.009	208.2 x 10 <sup>-7</sup>	20.92 x 10 <sup>-6</sup>	30.0 x 10 <sup>-3</sup>	29.9 x 10 <sup>-6</sup>	0.700
400	0.8711	1.014	230.1 x 10 <sup>-7</sup>	26.41 x 10 <sup>-6</sup>	33.8 x 10 <sup>-3</sup>	38.3 x 10 <sup>-6</sup>	0.690
450	0.7740	1.021	250.7 x 10 <sup>-7</sup>	32.39 x 10 <sup>-6</sup>	37.3 x 10 <sup>-3</sup>	47.2 x 10 <sup>-6</sup>	0.686
500	0.6964	1.030	270.1 x 10 <sup>-7</sup>	38.79 x 10 <sup>-6</sup>	40.7 x 10 <sup>-3</sup>	56.7 x 10 <sup>-6</sup>	0.684
550	0.6329	1.040	288.4 x 10 <sup>-7</sup>	45.57 x 10 <sup>-6</sup>	43.9 x 10 <sup>-3</sup>	66.7 x 10 <sup>-6</sup>	0.683
600	0.5804	1.051	305.8 x 10 <sup>-7</sup>	52.69 x 10 <sup>-6</sup>	46.9 x 10 <sup>-3</sup>	76.9 x 10 <sup>-6</sup>	0.685
650	0.5356	1.063	322.5 x 10 <sup>-7</sup>	60.21 x 10 <sup>-6</sup>	49.7 x 10 <sup>-3</sup>	87.3 x 10 <sup>-6</sup>	0.690
700	0.4975	1.075	338.8 x 10 <sup>-7</sup>	68.10 x 10 <sup>-6</sup>	52.4 x 10 <sup>-3</sup>	98.0 x 10 <sup>-6</sup>	0.695
750	0.4643	1.087	354.6 x 10 <sup>-7</sup>	76.37 x 10 <sup>-6</sup>	54.9 x 10 <sup>-3</sup>	109.0 x 10 <sup>-6</sup>	0.702
800	0.4354	1.099	369.8 x 10 <sup>-7</sup>	84.93 x 10 <sup>-6</sup>	57.3 x 10 <sup>-3</sup>	120.0 x 10 <sup>-6</sup>	0.709
850	0.4097	1.110	384.3 x 10 <sup>-7</sup>	93.80 x 10 <sup>-6</sup>	59.6 x 10 <sup>-3</sup>	131.0 x 10 <sup>-6</sup>	0.716
900	0.3868	1.121	398.1 x 10 <sup>-7</sup>	102.9 x 10 <sup>-6</sup>	62.0 x 10 <sup>-3</sup>	143.0 x 10 <sup>-6</sup>	0.720
950	0.3666	1.131	411.3 x 10 <sup>-7</sup>	112.2 x 10 <sup>-6</sup>	64.3 x 10 <sup>-3</sup>	155.0 x 10 <sup>-6</sup>	0.723
1000	0.3482	1.141	424.4 x 10 <sup>-7</sup>	121.9 x 10 <sup>-6</sup>	66.7 x 10 <sup>-3</sup>	168.0 x 10 <sup>-6</sup>	0.726

Formulas for interpolation; T in absolute temperature (K), and  $\sigma$  = standard deviation:

$$\rho = \frac{348.59}{T} \quad (\sigma = 9 \times 10^{-4}); \quad f(T) = A + BT + CT^2 + DT^3, \quad \text{where coefficients are given below:}$$

f(T)	A	B	C	D	$\sigma$
$c_p$	1.0507	-3.645 x 10 <sup>-4</sup>	8.388 x 10 <sup>-7</sup>	-3.848 x 10 <sup>-10</sup>	4 x 10 <sup>-4</sup>
$\mu \times 10^7$	13.554	0.6738	-3.808 x 10 <sup>-4</sup>	1.183 x 10 <sup>-7</sup>	0.4192
$\kappa \times 10^3$	-2.450	0.1130	-6.287 x 10 <sup>-5</sup>	1.891 x 10 <sup>-8</sup>	0.1198
$\alpha \times 10^6$	-11.064	7.04 x 10 <sup>-2</sup>	1.528 x 10 <sup>-4</sup>	-4.476 x 10 <sup>-8</sup>	0.4417
Pr	0.8650	-8.488 x 10 <sup>-4</sup>	-1.234 x 10 <sup>-6</sup>	-5.232 x 10 <sup>-10</sup>	1.623 x 10 <sup>-3</sup>

*Sutherland's law* for air viscosity:

$$\mu = \mu_0 \left( \frac{T}{T_0} \right)^{1.5} \left( \frac{T_0 + S}{T + S} \right)$$

where

- $\mu_0 = 1.71 \times 10^{-5}$  kg/(m s)
- $T_0 = 273.15$  K
- $S = 110.4$  K
- T must be in units of K.